### **ABSTRACT**

In support of NASA ARMD's code validation project, we have made significant progress by providing the first quantitative single-shot multi-scalar data from a turbulent elevated-pressure (5 atm), swirl-stabilized, lean direct injection (LDI) type research burner operating on CH4-air using a spatially-resolved pulsed-laser spontaneous Raman diagnostic technique. The Raman diagnostics apparatus and data analysis that we present here were developed over the past 6 years at Glenn Research Center. From the Raman scattering data, we produce spatiallymapped probability density functions (PDF's) of the instantaneous temperature, determined using a newly developed low-resolution effective rotational bandwidth (ERB) technique. The measured 3-scalar (triplet) correlations, between temperature, CH4, and O2 concentrations, as well as their PDF's, also provide a high-level of detail into the nature and extent of the turbulent mixing process and its impact on chemical reactions in a realistic gas turbine injector flame at elevated pressures. The multi-scalar triplet data set presented here provides a good validation case for CFD combustion codes to simulate by providing both average and statistical values for the 3 measured scalars.

# Single-Shot Scalar-Triplet Measurements in High-Pressure Swirl-Stabilized Flames for Combustion Code Validation

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HPGB Facility ERB/SE-5 H2-Air Flame at 20 atm

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## **Overview**

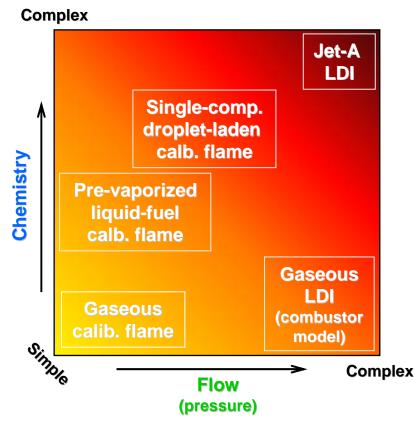


**Goal:** Provide critical chemical and physical data in turbulent combustion at realistic subsonic/supersonic cruise condition for *validating predictive low-emissions combustor codes* 

**Challenges:** Successful accurate multiscalar measurements (temperature, major species mole fractions, mixture fraction, velocity) in optically-harsh droplet-laden flows at *high pressures*.

## Strategy (tasks):

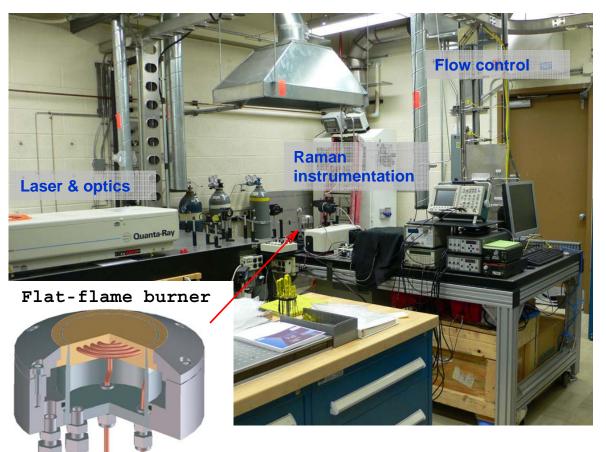
- ✓ Develop a quantitative time-resolved *laser Raman diagnostics*
- ✓ A series of experiments of increasing flow complexity: (i) Calibration burner (e.g., flat-flame burner), (ii) Realistic concept burners (swirl-stabilized burner) at elevated pressures up to 10~30 atm
- ✓ Address the effects of chemical complexity: H<sub>2</sub>, CH<sub>4</sub>, single-component liquid (e.g., hexane, iso-octane, n-heptane), jet-A fuel





## **Atmospheric Pressure Combustion Diagnostics Facility**





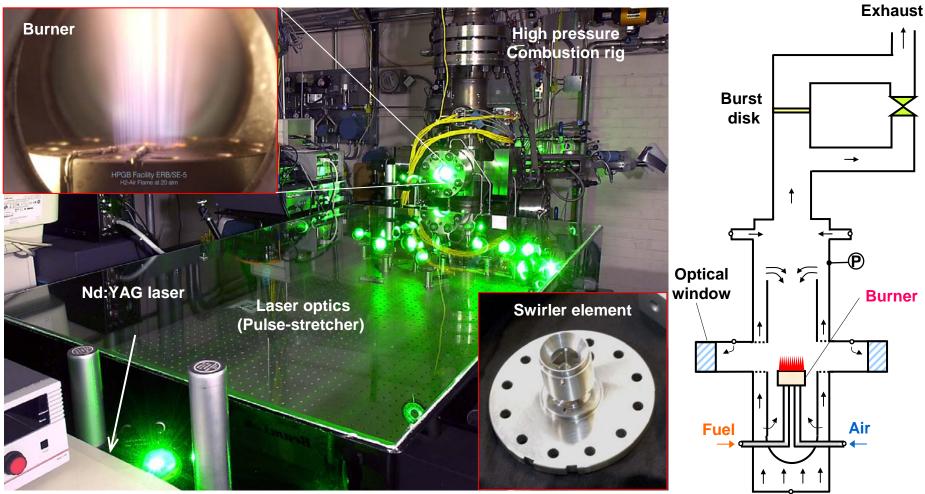
- Temperature and species reference
- Diagnostics calibration
- Gaseous fuels
- Pre-vaporized single-component liquid fuel (electronically controlled fuel vaporizer and heating system)
- Support visible and UV laser Raman systems
- Study on optical characteristics of sooting flames and liquid fuel combustion chemistry
- Support the Aeronautics milestones to mitigate risks due to PSO holding up high-pressure experiments

Fuel/air mixture (gas, or pre-vaporized liquid fuel)



## **High Pressure Burner Facility (SE-5)**



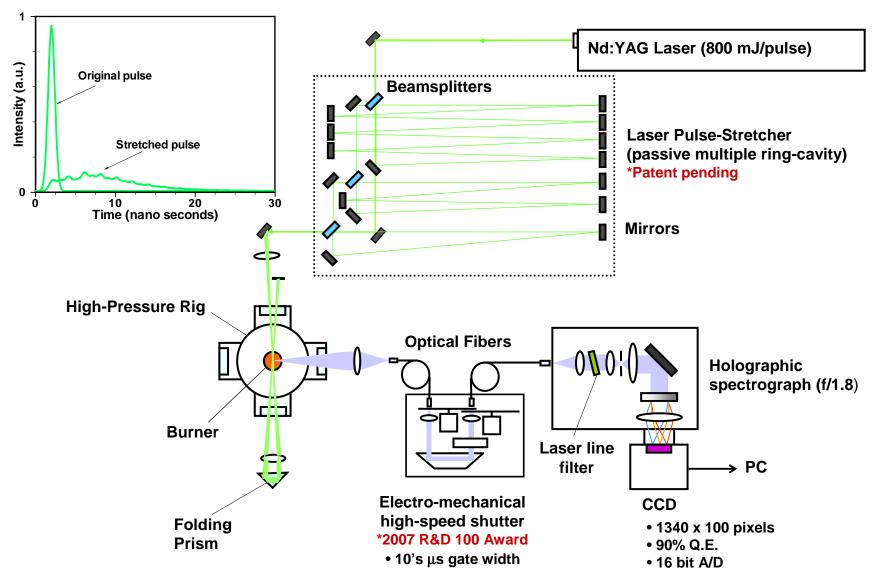


- √ Remotely controlled with auto process controller
- √ Pressure up to 30 atm (currently 10 atm requested for safety permit)
- ✓ Versatile for burner platform (calibration burner, turbulent jet, LDI)
- √ Optical access (4 ports)
- √ Gaseous and liquid fuel capacity
- ✓ Air pre-heater installed (up to 1200F)



## **Raman Scattering Diagnostic System**







## **Raman Spectra Observed in Combustion**

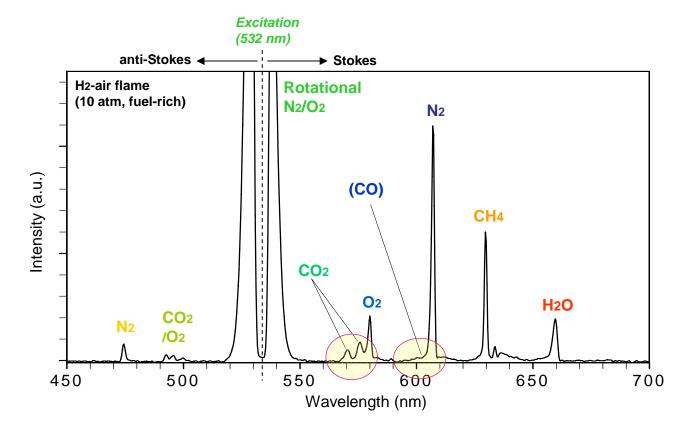


## **Advantage**

- Raman scattering –simultaneous multiple species concentration and temperature
- Time-average (mean) or Single-shot (instantaneous; rms)
- Quantitative

## **Challenge**

• Spectral interference ('Cross-talk')





## **Raman Calibration Experiments & Simulation**



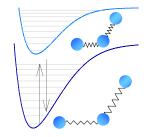
## <u>Platform</u>



Static cell



Calib. burner



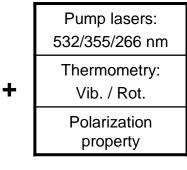
Raman spectrum simulation

## <u>Fuel</u> (lean ~ rich)

H2
H2-CO
CH4
Single-component liquid
(pre-vaporized)
Single-component liquid
Multi-component (jet-A)

Air

## **Raman Instrumentation**



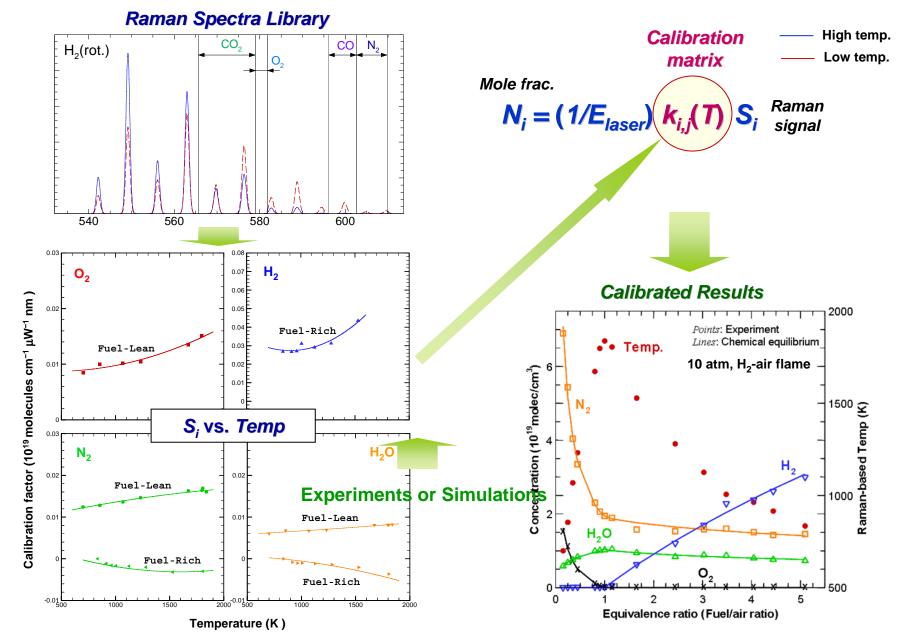
## Matrix elements, ki,j(T) (Cross-talks)

Major species (H2, O2, N2, CO, CO2, H2O, HC's)
H2 ►CO
H2 ►CO2
H2 ►H2O
N₂ ►CO
O2 ►CO2
CO2 ►N2
C2*►N2, CO, HC's
PAH ►Broadband background



## **Determine the 'Cross-talk' Calibration Matrix**



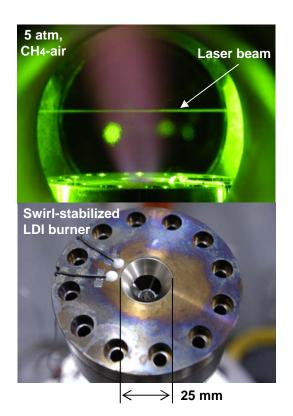




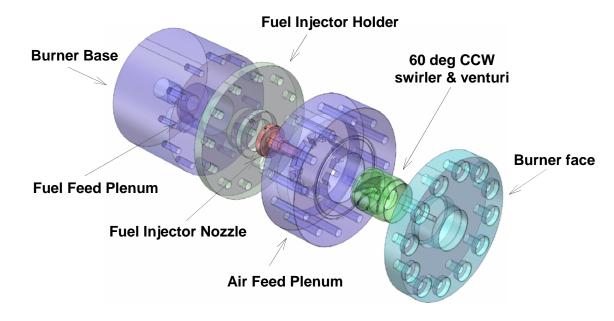
# NASA Lean Direct Injection (LDI) Swirl-Stabilized Research Burner — Preliminary Test —



## Photos of LDI flame and burner face



## Single Gaseous LDI Injector Design (Solid model for CFD)



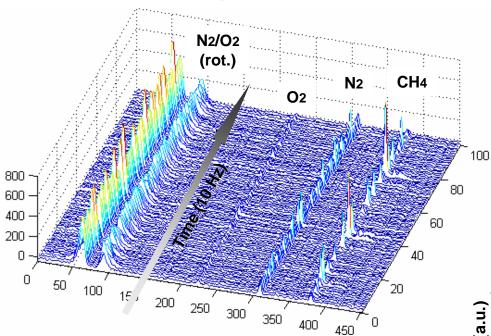
- Swirl-stabilized direct fuel injection design (gaseous)
- Integrated with existing high-pressure rig
- 6 jets (0.8 mm in dia.) angled at 45 deg to burner axis
- Initial test on H<sub>2</sub> and CH<sub>4</sub> fuels with unheated air at 5 atm
- Collaborating with National Combustor Code (NCC)



## **Single-Shot (Time-Resolved) Raman Measurements**



#### Time variation of single-shot Raman spectra in CH4-air LDI flame (5 atm, Phi = 0.5)

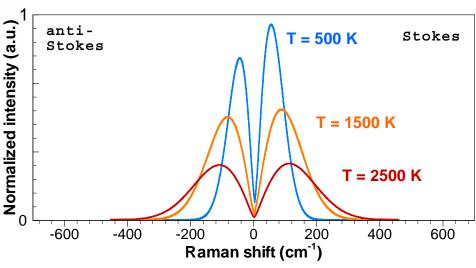


Raman shift (or wavelength)

- Temperature from the rotational spectra has been measured with accuracy 7%
- Data permits statistical PDF's of temperature and species
- Provides a signature for characterizing degree of mixing and reaction

- One shot = One instance (space-time point)
- Single-shot Raman data shows "random" change due to turbulence
- Direct and simultaneous measurement of fuel/oxidizer concentrations and temperature (with data processing with calibration matrix)
- Developed new thermometry approach (rot. bandwidth) with high SNR

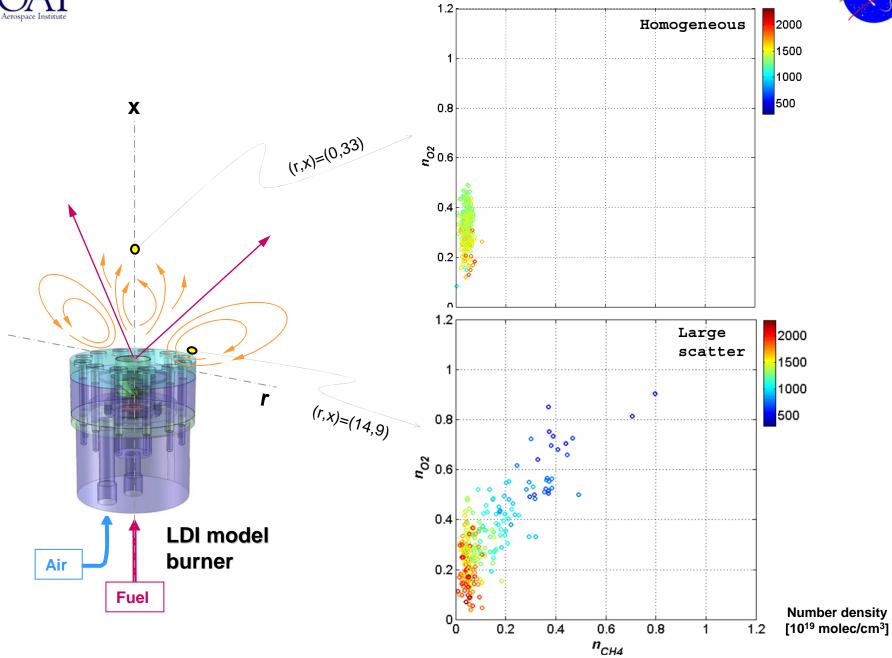
#### Rotational N2 Raman band





## **Processing the Single-Shot Data**

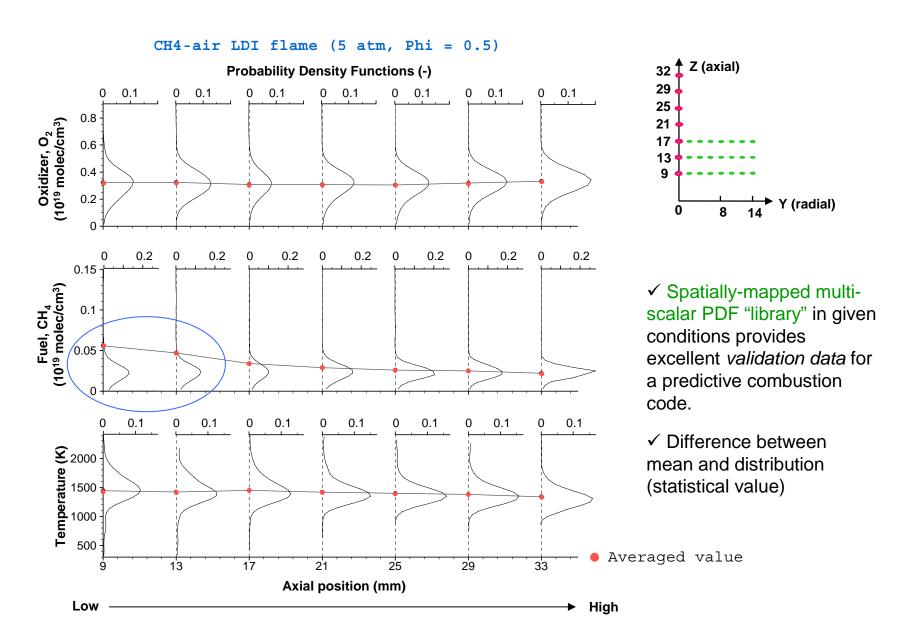






# **Probability Density Functions: PDF's** (Temperature, CH<sub>4</sub>, and O<sub>2</sub>)

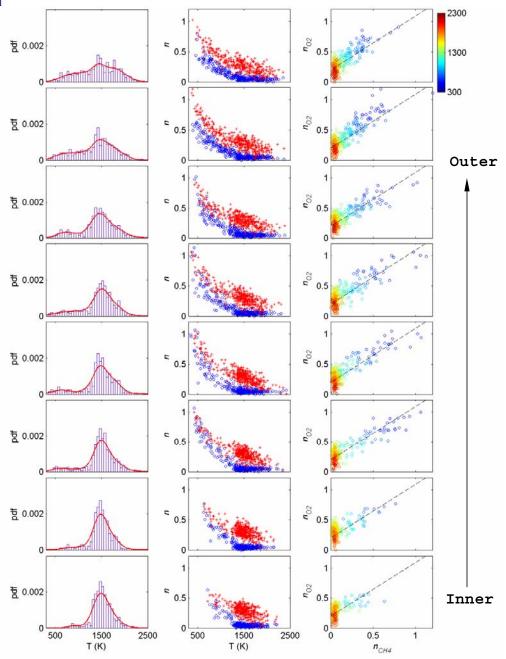


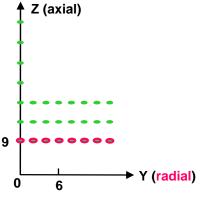




## Multiscalar Analysis in 5-atm CH<sub>4</sub>-Air LDI Flame







## *'Production-mode' data*: Direct output from a MATLAB code

- 1. Temperture PDF's
- 2. Temp. vs species correlations (*red*: O<sub>2</sub>; *blue*: CH<sub>4</sub>)
- 3. Triplet correlations (temp-ox-fuel) (with a global Phi of 0.5 line)

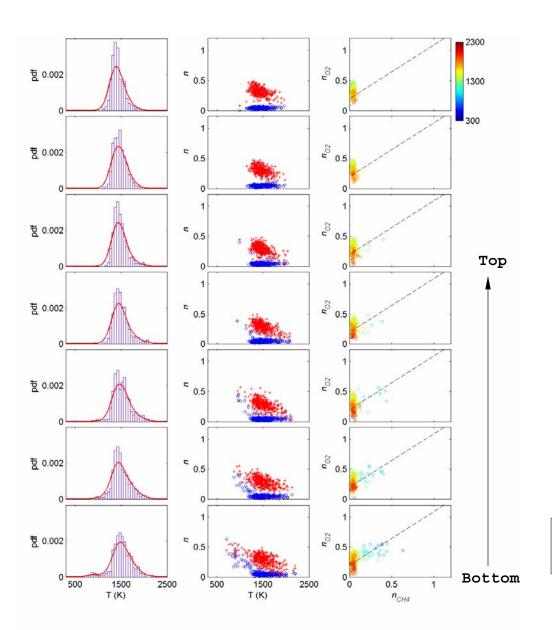
## Data interpretation

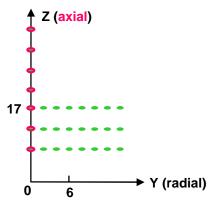
- ✓ Hot spots (high NOx)
- ✓ Unique bi-modal PDF's recirculation zone, or combustion oscillations
- ✓ Turbulent-chemistry interaction
- √ Fuel-air mixing characteristics
- ✓ Unburnt pockets



## Multiscalar Analysis in 5-atm CH<sub>4</sub>-Air LDI Flame







## Data interpretation

- ✓ Gaussian-like narrow Temp. distribution (centered around the adiabatic temp. at Phi of 0.5)
- √ No fuel (CH<sub>4</sub>) residual = fully consumed
- ✓ No hot or cold spots



✓ Homogeneous, well-reacted post-flame zone



## **Conclusions**



- A single-shot (time-resolve) capability of a laser Raman diagnostics has been confirmed.
- Preliminary data of time-resolved multiscalar data in a high-pressure (turbulent) swirl-stabilized flame has been acquired.
- A new single-shot data-processing scheme (computer code) has been developed for 'production mode' thermo-chemical analysis.
- Scalar PDF's and 3D (temp-oxy-fuel) correlations showed promising capability of future use in code validation

## **Work-in-progress**

- APCD (atmospheric pressure combustion diagnostics) facility is under construction to calibrate and improve the Raman diagnostics applicable to liquid fuels
- Modified visible and UV Raman diagnostic systems is under development to cope with harsh environments
- Computer code to simulate Raman spectra of major species including CO<sub>2</sub> and H<sub>2</sub>O (except hydrocarbon) is under development to complete the calibration matrix
- Pressure Safety Office safety permits, variances in process.